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5 **Epistemic uncertainties in structural modeling: a blind benchmark**
6 **for seismic assessment of slender masonry towers**

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75 **ABSTRACT:**
76

77 The paper reports the results of a blind benchmark developed as a part of the preliminary activity of
78 the research project RiSEM (Italian acronym for Seismic Risk on Monumental Buildings). The
79 benchmark was aimed at comparing the results obtained with different analytical models and/or
80 numerical analysis techniques (variational approach, finite elements, macro-elements, equivalent
81 frame, etc.) for the assessment of the nonlinear structural behavior of two cantilever masonry
82 elements with different slenderness under increasing horizontal loads. The analyzed elements were
83 characterized by a deliberately simple geometry, and the comparison between the numerical results
84 had a twofold purpose. On the one hand, it aimed at estimating the effects of the epistemic
85 uncertainties that are related to the different models and numerical techniques. On the other hand, it
86 aimed at reaching a proper evaluation of the influence of parameters describing the post-elastic
87 behavior of the structural typology analyzed within the research project (specifically, the masonry
88 towers). Both these objectives were necessary to further proceed with the development of simplified
89 numerical models needed for the subsequent risk analysis. For both slenderness, the results have
90 highlighted a significant dispersion of both the displacement capacity and post-peak softening
91 branch of the capacity curves. In addition, after some elaborations, it has been observed that the
92 dispersion of the results is proportional to both the shear-force and displacement level.
93
94

95 **KEYWORDS:**
96

97 Epistemic uncertainties; Modeling techniques; Pushover analysis; Seismic assessment; Slender
98 masonry towers; Unreinforced masonry.
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100
101

102 **Introduction**

103

104 In the field of Civil Engineering, the use of proper mechanical models and numerical codes is
105 frequently required to perform a broad range of tasks (*e.g.*, damage and/or structural identification,
106 response predictions to various loads, etc.) whose results constitute an important support input for
107 decision looking for structural problem solving. Usually solutions offered by the various modeling
108 techniques are affected by a certain degree of uncertainty, which asks for a proper quantification by
109 means of reliability analyses. Uncertainties affecting the physical systems can be grouped in two
110 main categories: aleatory and epistemic uncertainties (Helton and Oberkampf 2004). The first group
111 collects the uncertainties that are typically due to the randomness of the natural phenomena; these
112 uncertainties are unavoidable but today they can be efficiently approached within the framework of
113 probability theory (Lin 1967, Lin and Cai 1995). The second group considers the inaccuracy due to
114 the lack-of-knowledge. Several sources of inaccuracy can occur and, generally, epistemic
115 uncertainties collect a wide range of potential incomplete knowledge: the hypotheses underlying a
116 model (*i.e.* the ability of the model to describe the system of interest, together with the model
117 simplifications), the uncertainty in the model parameters, the observation errors, the uncertainty on
118 the software development, etc. (Der Kiureghian and Ditlevsen 2009).

119 While aleatory uncertainties are typically irreducible and non-subjective (since they are intrinsic
120 with the variations associated to the physical environment under consideration and/or with
121 uncertainties in resistance and other parameters of structural materials), the epistemic uncertainties
122 are (potentially) reducible since they are substantially due to ignorance or roughness in modeling
123 the overall physical environment. In last decades the distinction between aleatory and epistemic
124 uncertainties has become very significant. According to Pan et al. (2011), the factors that may
125 contribute to originate epistemic uncertainties can be classified as follows: 1) Vagueness
126 (information that is imprecisely defined, unclear, or indistinct); 2) Non specificity (presence of
127 several plausible alternatives); 3) Dissonance (existence of totally or partially conflicting evidence);
128 4) Ignorance. A general discussion on the classification of epistemic uncertainties is reported in Der
129 Kiureghian and Ditlevsen (2009), where the interested reader is referred.

130 The increasing interest of the scientific community toward the systematization of this class of
131 uncertainties, and their treatment, is demonstrated by the growing number of researches (Most
132 2011; Bradley 2010; Lagomarsino 2011; Tondelli et al. 2012) that spread the concept of epistemic
133 uncertainties in those fields of engineering where the effects of the lack-of-knowledge were (but
134 still are) traditionally assessed by heuristic sensitivity approaches. In Earthquake Engineering, the
135 sources of uncertainty due to the earthquakes characteristics (intensity and record-to-record

136 variability) are commonly classified as aleatory uncertainties; while the uncertainties on mechanical
137 parameters, constitutive models and cyclic behaviors are related to epistemic uncertainties.
138 Among all the potential sources of epistemic uncertainties, the paper aims to deepen the effects of
139 the so-called model framework uncertainties, *i.e.* those uncertainties that are due to the uncertainty
140 in the underlying science and algorithms of a numerical model or an analytical approach. This class
141 of epistemic uncertainty is fairly broad, and includes several sources of vagueness/ignorance (Der
142 Kiureghian and Ditlevsen 2009); among them: incomplete scientific data; lack-of-knowledge about
143 the factors that control the behavior of the system being modeled; the effects of the hypotheses
144 affecting the mechanical models; the effects deriving from the use of a model outside the
145 framework for which it was originally developed. This is, in fact, an actual challenge for the
146 scientific community since, with the advance of technology, modeling and simulation are
147 increasingly used in engineering science, and their complexity is correspondingly growing in order
148 to treat more sophisticated nonlinear physical processes. Nevertheless, more sophisticate modeling
149 requires more input variables to characterize the physical problem, this leading to additional greater
150 (epistemic) uncertainties. In addition, numerical models are usually tuned through comparison with
151 available experimental results (the models are fitted in order to reproduce a reduced number of
152 measurements identifying the unknown input parameters by an inverse strategy) but, at the same
153 time, they are employed to predict the structural behavior under exceptional loads, and predictions
154 for new extreme load cases may be inaccurate. This is true in general, but in case of the modeling of
155 masonry constructions the problem is amplified by the great, and growing, number of approaches
156 and numerical models proposed by the research community (Theodossopoulos and Sinha 2013).
157 Discussion of these uncertainties, and their effects, is herein approached through the results of a
158 blind benchmark on the seismic assessment of cantilever masonry beams. The benchmark was
159 developed within one of the research lines of the project “*RiSEM - Rischio Sismico negli Edifici*
160 *Monumentali*” (Seismic Risk on Monumental Buildings), a research project funded by the Tuscany
161 Regional Administration (Italy) that lasted from 2011 to 2013. The project aimed, through the
162 creation of a network gathering complementary expertise, at deepening the technical and scientific
163 knowledge in the field of the seismic assessment of monumental masonry buildings, with specific
164 reference to historic masonry towers. Taking into account the difficulties that arise in developing
165 exhaustive experimental campaigns in monumental building by traditional techniques, the project
166 aimed at developing expeditious methodologies to evaluate the main structural characteristics
167 needed for the subsequent seismic assessment at territorial scale.
168 The blind benchmark was focused at assessing the effects of model framework uncertainties on
169 those parameters that are of primarily relevance for the definition of the seismic vulnerability of this
170 typology of structures (herein idealized as a cantilever beam). The benchmark compared the results

171 offered by different analytical models and numerical analysis techniques (variational approach,
172 finite elements, macro-elements, equivalent frame, etc.) in the definition of the structural behavior
173 under horizontal loads of two cantilever masonry elements characterized by different slenderness.
174 The comparison between the results obtained by different models and computer codes has a twofold
175 purpose. On the one hand, the estimation of the effects of the epistemic uncertainties (which are
176 herein mainly related to the different analytical models and numerical analysis techniques adopted);
177 on the other hand, the definitions of the parameters which describe the crisis of the structural
178 elements (in order to further proceed with the development of simplified numerical models). The
179 aim, therefore, is not to calibrate the mechanical and numerical parameters in order to reproduce
180 experimental or reference results, but to evaluate (assuming a minimum common set of material
181 parameters) the potential errors made in the prediction of the nonlinear behavior of a simple
182 structural system using different numerical and/or analytical approaches.

183 The paper is organized as follows: in a first section the blind benchmark is described together with
184 the employed mechanical parameters, while the numerical models are briefly described in a
185 subsequent section. After, the results are summarized and critically compared; some conclusions are
186 reported at the end of the paper.

187

188 **Description of the blind benchmark**

189

190 The idea was to perform a blind benchmark among several research groups working in the field of
191 numerical modeling of the nonlinear structural behavior of masonry structures. Each researcher has
192 chosen the proper modeling technique, based on his own experience. The benchmark is a “blind”
193 one as no reference with existing data is done: “real” results are not known and they will never be
194 available, so every researcher has to do his/her best to guess which were the ultimate capacity of a
195 given structural elements without any proof of what he/she is assessing is correct. The same
196 reliability level has been given to each of the analyses, even if an “expert a priori judgment” could
197 have been assessed to those methods and analyses which are known not to be as accurate as others
198 in facing the proposed problem.

199

200 *Geometric characteristics and mechanical properties*

201 The request of the benchmark was the one of determining the ultimate load and displacement under
202 an increasing horizontal load of two simple cantilever masonry panels (Figure 1), characterized by
203 different slenderness. The first case study, in particular, was a purely theoretical panel with
204 dimensions 10 (B , width) \times 40 (H , height) \times 1 (t , thickness) meters; dimensions in the second case
205 were given as follows: 4 \times 40 \times 1 ($B \times H \times t$) meters. The two cantilever beams will be respectively

206 referred to as $[10 \times 40]$ and $[4 \times 40]$ in the following. Both panels are forced to undergo a plane
207 stress or strain state, that is they are mainly compressed and bent normally to their thickness. The
208 slendernesses of the case studies, $\lambda=H/B=4$ for the first case and $\lambda=10$ for the second one, were
209 selected in order to cover a wide enough range of slenderness of existing masonry tower according
210 to the goals of the research project RiSEM. Such slenderness may range, taking into account Italian
211 historic towers, between 3 (Pighin tower in Rovigo; Valente and Milani 2016) and 10 (Mangia
212 tower in Siena; Pieraccini et al. 2014). The differences in slenderness, moreover, allow analyzing
213 different failure modes (for bending or shear). Both panels were assumed fixed at the base.
214 The mechanical properties adopted in the models were assumed on the basis of some experimental
215 results obtained through in-situ shear-compression tests on a historic masonry (Galano and Vignoli
216 2001; Chiostrini et al. 2003) characterized by a chaotic texture made of stone or mixed stones and
217 bricks masonry; the thickness of the panels varied from about 300 mm to about 600 mm (Chiostrini
218 et al. 2003). Elastic parameters are reported in Table 1, together with the compressive and tensile
219 strength. The weight of the masonry was assumed equal to 18 kN/m^3 . Even if a suggested value for
220 Poisson's coefficient was reported as an input of the benchmark, in several models a null value was
221 used. The same was for the tensile and shear strength, assumed null in some models. Additional
222 physical parameters, when required by each modeling approach, are specified in the corresponding
223 section; they have been chosen and used independently by all the authors in their numerical
224 modeling.

225

226 *The performed analysis*

227 To model the seismic capacity of the panels, nonlinear static analyses with uniform load
228 distributions, monotonically increased up to failure, were performed. This procedure, now common
229 in the structural engineering practice, is used to obtain an estimation of the so-called capacity curve
230 of a structure (to be compared with the seismic demand), which represents a basic datum to predict
231 both strength and ductility of the structure under consideration. With the aim to compare the
232 different approaches, the capacity curves were evaluated, and the comparison was performed by
233 taking directly into account the pushover curves calculated by each code in term of dimensionless
234 shear, $\alpha = V/W$, being V the base shear and W the total weight of the beam. The ratio at the base of
235 the cantilever beam is hence plotted against the ratio between the horizontal displacement d of the
236 center of mass of the upper section of the beam and the beam height, H , that is by considering as a
237 displacement parameter the drift $\theta = d/H$ (Figure 1).

238

239

240

241 **Numerical models and analytical approaches**

242

243 The blind benchmark aims at quantifying the effects of the epistemic uncertainties associated with
244 the different modeling approaches (model framework uncertainties). To this end, the benchmark
245 tests consider a wide class of analytical models and numerical analysis techniques. A first class of
246 models includes numerical approaches specifically implemented to analyze the seismic response of
247 masonry buildings (3Muri, 3DMacro), and resorts in different level of assumptions and
248 simplification. A second class of numerical applications includes some general-purpose finite
249 element (FE) codes, distributed under both commercial (DIANA, ANSYS) and OpenSource (Code
250 ASTER) licenses. Each code adopts a different approach to model the nonlinear behavior of
251 masonry. An additional class of research numerical instruments has been moreover tested (NOSA-
252 ITACA, MADY, SMARTmasonry, VDM), specifically designed and developed within the
253 Authors' research groups to model the mechanical behavior of masonry constructions. Although the
254 selected codes were chosen with the aim to cover a wide and representative range of tools
255 commonly employed (or that can be employed) to reproduce the specific nonlinear behavior of
256 masonry and masonry structures, additional approaches are today available. Among them, it is
257 noteworthy, f.i., the combined finite discrete element method (FDEM) (Munjiza 2004). In addition,
258 with specific reference to the general purposes FE codes, worth noting are also, even if not
259 employed in this research, the codes ABAQUS (Tarque et al. 2014), ADINA (Bennati et al. 2005)
260 and LUSAS (Adam et al. 2010). To report an exhaustive review is almost impossible, and the
261 interested reader is referred to Theodossopoulos and Sinha 2013 and Asteris et al. 2015.
262 Depending on the considered numerical approach, the problem has been analyzed by using one or
263 two-dimensional elements (with the assumption of plane stress or strain problem, depending on the
264 model framework), in other cases by using three-dimensional brick elements, depending on the
265 characteristics of the specific code. An overview of such numerical methods, as well as a brief
266 description of their specific theoretical aspects, is provided in order to allow easy comparison.

267

268 ***3Muri***

269 The software 3Muri is a user-friendly computer code specifically proposed for the seismic analysis
270 of regular masonry buildings through pushover analyses. The code, originally developed at the
271 University of Genoa (Italy) and subsequently implemented in the commercial software 3Muri
272 (Cattari et al. 2004; Lagomarsino et al. 2013), uses an equivalent frame modeling (EFM) approach
273 according which the structure is idealized as a combination of one-dimensional macro-elements and
274 is analyzed as a framed structure. Therefore, each wall of the masonry structure is subdivided into

275 piers and spandrels, modeled with the one-dimensional macro-elements, which are connected by
276 rigid nodes. The in-plane behavior of the macro-elements, both piers and spandrels, is assumed as
277 elastic-perfectly plastic, with shear resistance and ultimate displacement obtained according the
278 provisions of the Italian Code (NTC2008, 2008). In particular, the ultimate shear resistance is
279 evaluated as the minimum between the resistance values for bending and diagonal cracking (in case
280 of existing structures); while the ultimate displacement is conventionally assumed as a percentage
281 of the height of the macro-element, considering the corresponding typology of collapse. The
282 software, being based on the EFM approach, needs a limited number of degrees of freedom (DOFs)
283 and it is hence possible to analyze large regular masonry structures with a reduced computational
284 effort. Herein some attempts were done in order to employ the code to analyze a cantilever masonry
285 beam. The application is hence interesting in order to evaluate the effects deriving from the use of a
286 model outside the framework for which it was originally developed.

287 The [10 × 40] cantilever beam as a first attempt was discretized by using only one macro-element
288 (Figure 2a). Due to the fact that the distribution of the horizontal load is composed of a single force
289 on the top of the macro-element, this discretization does not allow the reproduction of the uniform
290 load distribution. In order to obtain a uniform load distribution, a sensitivity analysis was hence
291 performed dividing the structure into several horizontal levels, starting from the initial single
292 macro-element (1×1) until 12 macro-elements (1×12) (Figure 2a - d). The pushover curves show a
293 convergence on both the collapse multiplier and the drift obtained with a 12 macro-elements
294 discretization, which allowed the reproduction of the uniform horizontal load distribution: the
295 collapse multiplier α converges to the value of 0.20, while the maximum drift θ is approximately
296 equal to 4%. Starting from the (1×12) discretization, a subsequent sensitivity analysis was carried
297 out in order to assess the influence of the slenderness of each macro-element, dividing them along
298 the base length. In the starting discretization (1×12 configuration), the macro-element dimensions
299 are 10.00 × 3.33 m, with slenderness $h_e/b_e=0.33$ (being h_e the height of the elements and b_e their
300 base). From this configuration, 2×12 (Figure 2e), 3×12 and 4×12 (Figure 2f) configurations were
301 analyzed. In the last configuration the size of each macro-element was 2.50 × 3.33 m, with a
302 slenderness $h_e/b_e=1.33$. The comparison of the pushover curves showed that both the collapse
303 multiplier and the initial stiffness (up to about the 60% of the maximum multiplier) are the same in
304 each discretization. On the contrary, a dispersion was observed with respect to the ultimate drift,
305 with a stabilization around the value of 8% with the model (4×12). This model was hence assumed
306 as representative of the cantilever beam: it provides a collapse multiplier equal to 0.193 and a
307 maximum displacement equal to 8.3% of the total height (Table 5). The first elements that collapse
308 are the ones at the ground level where the axial load decreases during the evolution of the analysis,
309 reducing the shear strength of the considered elements.

310 With respect to the $[4 \times 40]$ cantilever beam, as in the previous case, the division into 12 levels
311 along the height showed a convergence of the collapse multiplier to the value of 0.075, while the
312 maximum drift is approximately 9.5 ‰. Starting from this discretization (1×12 configuration), the
313 influence of the slenderness of the macro-elements on the behavior of the beam was again
314 investigated. In the first model (1×12 configuration) the dimensions of macro-element were
315 4.00×3.33 m, with slenderness $h_e/b_e=0.83$. Subsequently, 2×12, 3×12 and 4×12 configurations were
316 evaluated. In the last configuration the size of each macro-element was 1.00×3.33 m, with a
317 slenderness $h_e/b_e=3.33$. As in the previous case both the collapse multiplier and the initial stiffness
318 are the same in each discretization. The ultimate drift ranges from a value of 9.5‰ (1×12) up to a
319 value equal to 15.1‰ (2×12, 3×12, 4×12). In this case, the representative model was the one with
320 the 2×12 macro-elements discretization, where the elements dimension is 2.00×3.33 m with
321 slenderness equal to 1.67 (similar to the one of the beam $[10 \times 40]$). The representative (2×12)
322 model provides a collapse multiplier equal to 0.076 and a maximum displacement equal to 15.1‰
323 of the total height. Even in this case, the collapse occurs due to the excessive bending at the base:
324 the vertical load became low with respect to compressive strength, the horizontal loads originate
325 tensile flexural cracking at left corner, and the panel begins to behave as a nearly rigid body rotating
326 around the toe with rocking.

327

328 **3DMacro**

329 The code 3DMacro, originally developed at the University of Catania (Italy) and subsequently
330 implemented in the commercial software 3DMacro, allows the analysis of the seismic behavior of
331 masonry buildings by using a two-dimensional macro-elements approach (Caliò et al. 2005). The
332 macro-element, in its first version, was a plane pinned quadrilateral element built with four rigid
333 edges. Two diagonal springs connect two opposite corners to simulate the masonry wall shear
334 behavior. Additional discrete distributions of nonlinear springs, with limited tension strength, are
335 employed to connect the rigid edges of neighboring macro-elements to simulate its interaction.
336 Springs normal to the sides of the macro-elements are introduced to simulate the axial and bending
337 deformability and to account for the crushing and flexural collapses, while parallel springs simulate
338 the sliding along macro-elements. The whole set of springs allows to properly simulate the
339 nonlinear in-plane behavior of masonry buildings through the effective reproduction of the main in-
340 plane collapse mechanisms of the masonry (flexural, shear-diagonal and shear-sliding failure). To
341 account for the out-of-plane collapse behavior of masonry walls the plane macro-element was
342 recently enriched by introducing a third dimension with additional nonlinear springs and additional
343 DOFs (Caliò et al. 2008; Caliò et al. 2012). Both the plane two-dimensional macro-element and the
344 enriched three-dimensional one allow a discrete equivalent representation of a masonry structure by

345 assembling the macro-elements: a masonry panel, considering its dimensions, can be modeled with
346 a unique macro-element or with a mesh of elements. Because of the reduced number of DOFs (in
347 case of meshing with two-dimensional macro-element, each panels has 4 DOFs) the approach
348 requires a suitable computational demand. Furthermore, only few parameters are required to
349 characterize the nonlinear behavior of the masonry material: the elastic E_w and shear G_w modulus,
350 the masonry compressive strength f_{wc} and the characteristic shear strength of the masonry τ_k .
351 Additional parameters required by the code are the distance between the interface nonlinear springs
352 and their ductility (both in tension and compression). Through the code, the seismic behavior of a
353 masonry building is evaluated by performing pushover analyses.
354 As a first step, in order to reproduce the uniform distribution of the lateral load, a discretization
355 along the height of the cantilever beams was performed. A parametric analysis was conducted by
356 varying the height of the macro-elements and additional analyses were performed to investigate the
357 influence of the horizontal dimension of the macro-elements and of the parameters defining its
358 behavior (such as the distance between the interface springs). It was verified that the analyzed
359 results, such as the collapse multiplier of the lateral load as well as the ultimate horizontal drift, are
360 not significantly influenced by the variations of the investigated parameters (*i.e.* both the pushover
361 curve and the collapse mechanism do not change). As a final mesh, the discretization of cantilever
362 beam in macro-elements involves the use of 16 macro-elements (dimensions 5×5 m) for the [10 ×
363 40] cantilever beam (Figure 3c) and 8 macro-elements (dimensions 4×5 m) for the [4×40] cantilever
364 beam. As a second step, several parametric analyses were performed to assess the stability of the
365 results with respect to the spacing between the interface nonlinear springs and their ductility. After
366 the test, in both cases, the distance between the interface nonlinear springs was assumed equal to 25
367 cm (*i.e.* about 20 spring each side). Suggested default values were maintained for the spring
368 ductility. The model of the cantilever beam [10 × 40] provides a collapse multiplier equal to 0.212
369 and an ultimate drift equal to about 4 ‰ (Table 5). For the model of the cantilever beam [4 × 40]
370 the collapse multiplier is about 0.076 and the ultimate drift about 10‰ (Table 6). In both cases, the
371 collapse occurs due to a flexural mechanism at the basement; both panels develop tensile cracks in
372 the interfaces. Figure 3 shows the collapse mechanism for the beam [10 × 40].

373

374 **DIANA**

375 The commercial FE code DIANA was used to model the two slender cantilever masonry beams
376 using three-dimensional (3D) 6-node isoparametric wedge elements (*TP18L*, Figure 4a). These
377 elements were selected to avoid mesh dependence of the cracks. After meshing, the final 3D
378 numerical model of the first cantilever beam [10 × 40] (Figure 4b) consisted of 1,804 nodes, 2,400
379 3D *TP18L* elements, corresponding to 5,280 DOFs. The final 3D numerical model of the second

380 cantilever beam [4×40] consisted of 820 nodes, 960 3D *TP18L* elements, for a total of 2,400
381 DOFs. Displacements of the base nodes were fixed and, in addition, displacements of nodes in the
382 transversal direction were also set equal to zero to analyze the plane strain state problem.
383 In order to model the cracking/crushing behavior, two models based on total strain (stress is defined
384 as a function of strain) are implemented in DIANA. The first model is the Total Strain Crack
385 Rotating (TSCR) where the stress-strain relationship is evaluated in the principle direction of the
386 strain vector. The second one is the Total Strain Crack Fixed (TSCF), where the stress-strain
387 relationship is evaluated in a fixed coordinate system which does not change once cracking is
388 initiated. Both models are smeared-crack models according to which the localized cracking
389 phenomenon is simulated in a disseminated way (taking advantage of the mesh-assembly of the FE
390 model in order to facilitate numeric computation) and both models allow the possibility of forming
391 two orthogonal cracks in each integration point. In this study the TSCF was employed since it has
392 been shown that this model is more appropriate for most engineering purposes (Lourenço et al.
393 1998). The compressive uniaxial behavior is characterized by a linear stress-strain relation until
394 about one third of the compressive strength, followed by a parabolic relation for the hardening
395 regime until reaching the compressive strength and another parabolic branch for the post-peak
396 softening according the Thorenfeldt model (Thorenfeldt et al. 1987). This model was chosen since
397 its formulation does not depend on the fracture energy and only the compressive strength and the
398 Young's modulus are required. The tensile uniaxial behavior is modeled according the linear
399 tension softening model of Hordijk with exponential softening behavior in tension (Hordijk, 1991)
400 assuming a constant shear retention factor β . This factor accounts for the residual strength (or
401 friction) between the two surfaces of a crack.
402 The values of the inelastic parameters required by the models were selected to reproduce the
403 uniaxial compressive and tensile strengths reported in Table 1, and are summarized in Table 2.
404 The fracture energy (corresponding to the integral of the stress-displacement diagram for uniaxial
405 stress and equating the energy needed to create a unit area of a fully developed crack) was estimated
406 by:

$$G_f = \frac{f_{wt}^2 \cdot h}{0.739 \cdot E_w} = 38.97 \text{ N/m} \quad (1)$$

407
408 where $h = \sqrt[3]{V} = 0.75 \text{ m}$ denotes the crack bandwidth and V is the volume of the element. It is
409 noteworthy that, using a smeared-crack model, the fracture energy must be normalized according to
410 an equivalent length h in order to obtain mesh-objective results with respect to the mesh refinement.
411 The crack bandwidth h depends on the elements type, size, shape, integration scheme, etc. For the

412 models employed within this study, DIANA assumes the default crack bandwidth h to be the cubic
413 root of the volume (all of the elements are solid elements).

414 The model of the cantilever beam [10 × 40] provides a collapse multiplier equal to 0.195 and an
415 ultimate drift equal to about 8 ‰ (Table 5). For the model of the cantilever beam [4 × 40] the
416 collapse multiplier is about 0.084 and the ultimate drift about 22.5‰ (Table 6). The cracking
417 pattern and vertical stresses corresponding to the maximum base shear obtained with the [10 × 40]
418 cantilever masonry beam are reported in Figure 4c.

419

420 ***Code ASTER***

421 The two slender cantilever masonry beams were also modeled by means of Code ASTER (acronym
422 of “Analyses des Structures et Thermo-mécanique pour des Études et des Recherches”), an Open
423 Source FE solver employed for numerical simulations of materials and structures developed at the
424 department “Analyses Mécaniques et Acoustiques” of Électricité de France (EDF). The
425 implementation of the code began in 1989, to meet the internal needs of EDF in the nuclear
426 industry, and it was released under the terms of the GNU GPL license in 2001. Under the Linux
427 operative system, Code ASTER is directly integrated with the platform Salome-Meca (a multi-
428 purpose platform for Pre- and Post-Processing for numerical simulation). The validation of the code
429 was extremely careful, many comparisons with experimental results and benchmarks with other
430 codes have been carried out by independent bodies by EDF. The code is particularly robust,
431 containing about 1,500,000 lines of code (written both in Fortran and in Python).

432 Through the code the two cantilever beams were modeled by using three-dimensional (3D) 8-node
433 isoparametric elements and, after meshing, the final 3D numerical model of the first cantilever
434 beam [10 × 40] (Figure 5a) consisted of 11,907 nodes, 9,600 3D elements, totalling 23,400 DOFs.
435 The final 3D numerical model of the second cantilever beam [4 × 40] (Figure 5b) consisted of 6,237
436 nodes, 4,800 3D elements, corresponding to about 12,250 DOFs. Displacements of the base nodes
437 were fixed, and displacements of nodes in the transversal direction were zeroed to account for a
438 plane strain state problem.

439 The concrete damage model of Mazars (Mazars and Pijaudier-Cabot 1989) was employed to model
440 the masonry nonlinear behavior. The model refers to the continuum damage mechanics, according
441 to the progressive degradation of material stiffness, due to the propagation of micro-cracks, is
442 described through a continuous approach. The damage model of Mazars is an isotropic scalar
443 damage model that assumes the damaged stiffness tensor as a scalar multiple of the initial elastic
444 stiffness tensor. The damage is characterized by a single scalar variable, the damage index, ranging
445 between 0 (no damage) and 1 (complete loss of strength). Due to the assumption of isotropic
446 behavior, the stiffness degradation in different directions decrease proportionally and is independent

447 of the loading direction. In addition, being a single scalar damage index model, it assumes that the
448 Poisson's ratio is not affected by damage (the relative reduction of all the stiffness coefficients is
449 the same). Strictly speaking, isotropic Mazars constitutive model could not be satisfactory when
450 used for modeling masonry structures, where orthotropic or anisotropic modeling should be used.
451 Nevertheless, it has to be observed that in many practical applications this approach is normally
452 retained as acceptable despite the level of approximation.

453 Mazars damage evolution law is expressed in an explicit form, relating damage parameter and
454 scalar measure of largest reached strain level in material, taking into account the principle of
455 preserving of fracture energy G_f . The model of Mazars is implemented in the code in two versions.
456 The first is a local approach where the stress at a point depends only on the deformation in the same
457 point. The second is a nonlocal version where local stresses depend not only on the deformation of
458 that point, but on the average strain defined in the neighborhood of the point. In this study the local
459 version was employed, being less computational demanding. Hence a preliminary sensitivity
460 analysis on the mesh size was performed to avoid stagnation of the results toward non-physical
461 solution (*i.e.* objectivity of the results with respect to the finite element mesh was investigated).
462 Apart the two elastic parameters (the Young's modulus and the Poisson's coefficient) the definition
463 of the model of Mazars requires 6 additional parameters, as reported in Table 3. The values of the
464 parameters were selected reproducing a uniaxial test on a cube with a material with the compression
465 and tension strength reported in Table 1. It is worth noting that, with this model, cracks at a
466 microscopic point have no particular direction and a macroscopic crack is then defined as the locus
467 of damage points. In fact, one of the advantages of such a model is the independence of the analysis
468 with respect to cracking directions, which can be simply identified a posteriori once the nonlinear
469 solution is obtained.

470 The damage map at collapse is reported (both for the $[10 \times 40]$ and the $[4 \times 40]$ cantilever masonry
471 beams) in Figure 5, while the principal compressive stresses are reported in Figure 6.

472

473 **ANSYS**

474 The commercial FE code ANSYS was used to model the two masonry beams by means of three-
475 dimensional 8-nodes isoparametric finite elements. The classical smeared-crack approach was
476 employed and the mechanical nonlinear behavior of masonry was modeled via two approaches.
477 In a first case the Willam-Warnke (WW) failure criterion was employed (Willam and Warnke
478 1975). This failure criterion, initially adopted for concrete, accounts for both cracking and crushing
479 failure modes through a smeared model. Despite the needing for five constants to define the
480 criterion, in most practical cases (thereby when the hydrostatic stress is limited by $\sqrt{3} f_c$) the adopted
481 failure surface is specified by means of only two constants: F_t and F_c (the uniaxial tensile and

482 compressive strength respectively). A shear transfer coefficient β is introduced (depending on the
483 crack status: open - β_t - or re-closed - β_c -) to take into account a shear strength reduction factor for
484 those subsequent loads inducing sliding (shear) across the crack face.

485 In a second case the Willam-Warneke failure criterion was combined with the Drucker-Prager
486 plasticity criterion (DP) originally proposed for geo-materials (Drucker and Prager 1952). In this
487 case, as a result, the material behaves as an isotropic medium with plastic deformation, cracking and
488 crushing capabilities. The material parameters required to define the model, the cohesion c and the
489 internal angle of friction φ , are introduced in such a way that the circular cone yield surface of the
490 DP model corresponds to the outer vertex of the hexagonal Mohr-Coulomb yield surface. The
491 constitutive parameters used for the DP criterion and the WW failure domain are reported in Table
492 4 (Model 1 combines the WW failure criterion with the DP plasticity one, while Model 2 adopts the
493 WW failure criterion alone). It is noteworthy to highlight the difference of the tensile and
494 compressive strengths of the DP criterion ($f_{tDP} = 0.34 \text{ N/mm}^2, f_{cDP} = 5 \text{ N/mm}^2$) and those of the
495 WW failure criterion ($F_t = 0.24 \text{ N/mm}^2, F_c = 6.0 \text{ N/mm}^2$). The combination of these parameters
496 allows for an elastic-brittle behavior in case of biaxial tensile stresses or biaxial tensile-compressive
497 stresses with low compression level. On the contrary, the material is elastoplastic in case of biaxial
498 compressive stresses or biaxial tensile-compressive stresses with high compression level (Betti et al.
499 2016).

500 The load control Newton-Raphson method was selected to solve the nonlinear equations and the
501 analyses were carried out by assuming a plane strain state; the Poisson's coefficient was assumed
502 equal to zero. Analyses were eventually conducted with and without geometric non linearities.
503 Preliminary tests were conducted to estimate the optimal mesh size, and the adopted size was
504 $0.5 \times 0.5 \times 0.5 \text{ m}$. Consequently, after meshing, the final 3D numerical model of the first cantilever
505 beam [10×40] consisted of 5,103 nodes, 401 3D *Solid65* elements, corresponding to 15,120 DOFs.
506 The final 3D numerical model of the second cantilever beam [4×40] consisted of 2,187 nodes,
507 1,280 3D *Solid65* elements, for a total of 6,470 DOFs. Increasing the number of finite elements did
508 not lead to any variation of results.

509 The cracking pattern obtained at collapse is reported, both for the [10×40] and the [4×40]
510 cantilever masonry beam in Figure 7, while the principal compressive stresses are reported in
511 Figure 8.

512

513 **NOSA – ITACA**

514 The FE code NOSA (acronym of “NOn-Linear Structural Analysis”) has been developing since the
515 1980s, by the Mechanics of Materials and Structures Laboratory of ISTI – CNR in Pisa, Italy
516 (Lucchesi et al. 2008), with the aim of testing new constitutive models for material. Within the

517 code, masonry is modeled as a homogeneous nonlinear elastic material with zero tensile strength
518 and either infinite or finite compressive strength (Del Piero 1989; Di Pasquale 1992) according to
519 the framework of no-tension (masonry-like) materials. The NOSA code has been successfully
520 applied to the static analysis of several historical masonry buildings (Lucchesi et al. 2008; Girardi et
521 al. 2015) and more recently to the seismic and dynamic analysis of masonry towers, beams and
522 domes (Binante et al. 2012).

523 The constitutive model and numerical techniques for solving equilibrium problems of masonry
524 constructions implemented in NOSA are described in Lucchesi et al. (2008). Briefly, masonry is
525 modeled as a nonlinear hyperelastic material with Young's modulus, $E > 0$, Poisson ratio ν (where
526 $0 < \nu < 1/2$), zero tensile strength and maximum compressive stress $\sigma_0 < 0$. Sym indicates the
527 vector space of symmetric tensors, while Sym^- and Sym^+ stand for the subsets of Sym constituted by
528 the negative and positive semidefinite tensors, respectively. It is assumed that the infinitesimal
529 strain, $\mathbf{E} \in Sym$, is the sum of an elastic part, $\mathbf{E}^e \in Sym$, and two mutually orthogonal inelastic
530 parts, $\mathbf{E}^f \in Sym^+$ and $\mathbf{E}^c \in Sym^-$, respectively called fracture strain and crushing strain. It is
531 moreover assumed that the Cauchy stress \mathbf{T} depends linearly and isotropically on \mathbf{E}^e . Finally, some
532 orthogonality conditions are imposed among tensors, to describe the elastic behavior of the
533 material, which cracks and crushes without dissipating energy. By exploiting the coaxiality of \mathbf{E} , \mathbf{T} ,
534 \mathbf{E}^f and \mathbf{E}^c , the stress tensor \mathbf{T} satisfying the constitutive equation can be expressed as a nonlinear
535 function of the total strain \mathbf{E} . The explicit expression for $\mathbf{T}(\mathbf{E})$ can be found in Lucchesi et al.
536 (2008), together with its derivative with respect to \mathbf{E} . These expressions are then implemented in
537 the Newton-Raphson scheme for solving the nonlinear algebraic system derived from discretisation
538 of the equilibrium problem.

539 In recent years NOSA code has been further enhanced within the framework of the NOSA –
540 ITACA project (www.nosaitaca.it/en) by integrating the NOSA code with the Open Source
541 platform Salome-Meca (employed to provide the pre-post processing environments for defining
542 geometries and visualizing results). The NOSA code has been substantially modified and equipped
543 with new finite elements, thus extending its application capabilities. An efficient implementation of
544 numerical methods for constrained eigenvalue problems, which enables conducting modal structural
545 analyses while taking into account the features of master-slave constraints (tying or multipoint
546 constraints), has been also embedded in NOSA (Porcelli et al. 2015).

547 The two slender masonry structures in the benchmark test have both been modeled via the NOSA –
548 ITACA code using eight-node plane stress elements. The assumed mechanical properties are those
549 reported in Table 1, except for the uniaxial tensile strength f_{wt} , which has set equal to zero. For the
550 beam $[4 \times 40]$, a mesh composed of 640 elements has been used, while 1600 elements were used to

551 discretize the $[10 \times 40]$ beam. After the dead load has been applied, the uniform lateral load is
552 subsequently increased incrementally and the nonlinear equilibrium equations are solved by means
553 of the Newton-Raphson scheme. It has been shown in Lucchesi et al. (2008) that the solution to the
554 equilibrium problem of masonry-like materials is unique in terms of stress, while special attention is
555 required when computing the displacements field. The results of the present pushover analyses have
556 been carefully tested for different meshes, element types and load histories; the analyses were
557 continued until numerical stability of the results was guaranteed. Figure 9 shows the distribution of
558 the σ_{zz} component of the stress tensor at the final step in the two cases.

559 Figure 10 and Figure 11 instead show the ε_{xx}^f and ε_{zz}^f components of the fracture strain tensor, which
560 reveal the distribution of cracked material in the structures and, lastly, Figure 12 shows the
561 distribution of the isostatic lines in the structures at collapse. The limit compressive strength is
562 attained in only a small portion of the structures. However, the high values reached by the fracture
563 strains, as well as their distribution in the structures, reveal the presence of a large triangular shaped
564 portion of masonry that is almost inactive, starting at about one-third the panels height on the
565 loaded side and reaching the opposite right corner. The high values of the fracture strain ε_{xx}^f on the
566 panel right side are induced by the concentration of vertical stresses in masonry, in the absence of
567 horizontal restraints. These phenomena are also confirmed by the distribution of the isostatic lines
568 (Figure 12).

569

570 **MADY**

571 MADY is a non-commercial code developed at the University of Florence (Italy) to perform
572 nonlinear (static and dynamic) analyses of masonry structures with predominantly flexural
573 behavior. It relies in a finite element discretization of the structures that are modeled through one-
574 dimensional elements. To describe the constituent masonry, the nonlinear elastic constitutive
575 equation for beams presented in Lucchesi and Pintucchi (2007) and Pintucchi and Zani (2009) is
576 used. It is developed in terms of generalized stress and strain, accounting for the axial stress
577 component alone and under the Euler-Bernoulli hypothesis. The material is assumed unable to
578 withstand tensile stresses and with limited compressive strength. Specifically, the model has been
579 developed for both solid and hollow rectangular cross-sections in order to study masonry arches as
580 well as free-standing masonry towers.

581 The FE procedure used to solve the equations of flexural and axial equilibrium, which in the
582 nonlinear range become coupled, make use of finite beam elements with three DOFs at each node:
583 axial and transverse displacements, plus rotation. The transverse displacement and rotation are
584 approximated using cubic interpolation, i.e. Hermite shape functions, which guarantee the
585 continuity of both of them, while linear shape functions are adopted for the axial displacement.

586 To obtain the solution, standard numerical procedures – the Newton-Raphson method and the
587 Newmark one for dynamic problems – are then used, which however require defining the element
588 stiffness matrix. The geometric nonlinearity can also be accounted for, as detailed in Pintucchi and
589 Zani (2009).

590 The model enables to perform static, pushover and nonlinear dynamic analysis with a minimum of
591 computational effort (Pintucchi and Zani, 2014). Moreover, for these one-dimensional masonry
592 structures, the uniqueness of the solution to static and dynamic problems has been proved in
593 Lucchesi et al. (2012) and Lucchesi et al. (2015) respectively. Enhancements of the model can also
594 be found in Pintucchi and Zani (2016).

595 The two cantilever beams have been analyzed with MADY using 45 one-dimensional elements;
596 increasing the number of finite elements did not lead to any significant variation of results. In the
597 first step of the analyses the vertical loads are applied, then the lateral loads are added
598 incrementally. Herein, the geometric nonlinearity has not been accounted for. The distribution of
599 the axial stress at collapse is reported (both for the $[10 \times 40]$ and the $[4 \times 40]$ cantilever masonry
600 beams) in Figure 13, while the obtained distribution of the damage is showed in Figure 14.

601

602 *SMARTmasonry*

603 SMARTmasonry is a code specifically developed at the University of Florence (Italy) to perform
604 hybrid modelling of masonry structures, allowing the mixing in the same structural model of
605 Discrete Elements and Finite Element discretized continuum. The latter can be possibly related to a
606 certain micro-structural modelling (Salvatori and Spinelli, 2010). In the proposed modelling for the
607 masonry cantilever, only rigid blocks put one on top of the other are used. The tower is discretized
608 in 20 rigid blocks, interacting through nonlinear interfaces. The models and their deformed
609 configuration at failure are reported in Figure 15a for the $[10 \times 40]$ cantilever beam and Figure 15b
610 for the $[4 \times 40]$ one. More refined discretizations result in no appreciable improvements in the
611 output quantities herein taken into consideration (first flexural bending mode and pushover capacity
612 curve).

613 Kinematic effects related to the displacements of centres of gravity of each single block have been
614 included in the co-rotational adopted model (large displacements and small deformation of the
615 interfaces between blocks). The constitutive model for the interface is nonlinear elastic in the
616 normal direction (with a limited compressive strength and a null tensile resistance), while it follows
617 an elastic - perfectly plastic law along the tangential direction (according to a Mohr-Coulomb
618 yielding criterion and non-associated flow rule). Damage in compression is related to the normal
619 strain level (corresponding to a volume-specific fracture energy). As a matter of fact, in the present
620 analyses the friction is large enough for not having any slip movement in the investigated structures.

621 In a first stage of the loading process, the self-weight is incrementally applied through a non-linear
622 load-controlled static procedure. In the final configuration, the tangent stiffness matrix is evaluated
623 and the modal analysis is performed. Then a nonlinear horizontal pushover analysis is performed,
624 by using an indirect displacement-controlled static procedure, where the displacement at the top of
625 the cantilever beam is evaluated by taking into account also the rotational contribution of the
626 topmost block. The same model has been also used to investigate effects of material uncertainties
627 (Salvatori et al. 2015) and of record-to-record variability in case of incremental nonlinear dynamic
628 analyses (Marra et al. 2016).

629 The analyses are carried out with and without compression damage. The latter case is useful to
630 highlight the nonlinear geometric effects (especially in the slenderer $[4 \times 40]$ structure).

631 Among the parameters previously described, the tensile strength has not been used, while a friction
632 coefficient 0.4 is assumed. In the analyses where the damage is included, an ultimate strain ε_u ,
633 corresponding to a value of compressive strain ductility $\mu_\varepsilon = \varepsilon_u/\varepsilon_e = \varepsilon_u E/f_{wc} = 2.0$, is considered.
634 This is a realistic value for historical masonry and corresponds to a volume-specific fracture energy
635 in compression $G_{cV} = 25.0 \text{ kJ/m}^3$.

636 The capacity curves reported in Figure 18 and Figure 19 follow the same path in the models with
637 and without damage. Of course, in presence of damage the capacity curves terminate at smaller
638 values of the top displacement. All curves show a progressive stiffness reduction due to the
639 reduction of the resisting section and progressive compression crushing. When compressive failure
640 does not occur earlier, after a peak where maximum load capacity is attained, a “softening” branch
641 arises due to geometrically nonlinear effects (this is particularly evident for the slenderer $[4 \times 40]$
642 structure, in absence of damage; Figure 19). When damage is not included the conventional ultimate
643 displacement is assumed correspondingly to a reduction of the base shear of 15% with respect to the
644 peak. This non-realistic condition could be reached if the strain ductility were $\mu_\varepsilon = 90.1$ for the $[10$
645 $\times 40]$ structure and $\mu_\varepsilon = 7.8$ for the $[4 \times 40]$ one.

646

647 ***Variational Damage Model (VDM)***

648 The variational approach to fracture was firstly proposed by Francfort and Marigo in their
649 pioneering paper (Francfort and Marigo 1998). They supposed that the formation and propagation
650 of cracks in brittle materials are governed by a minimization problem where the energy functional is
651 given by the sum of a bulk term and a Griffith’s fracture term (linearly proportional to the area of
652 the fracture surface). Later on, a variational approximation of the free-discontinuity problem
653 (Francfort and Marigo 1998) was proposed in (Bourdin et al. 2000), essentially to bypass the
654 numerical difficulties in representing discontinuous fields. The energy of the approximated problem
655 is a two-field functional, depending on the displacement field $\mathbf{u}(\mathbf{x}): \Omega \rightarrow \mathbb{R}^3$, and on the scalar field

656 $s(\mathbf{x}): \Omega \rightarrow [0,1]$, which represents a damage parameter, with $s = 1$ for sound material and $s = 0$ for
657 fractured material. In an evolution process, s can only decrease, thus avoiding material healing.
658 The form of the functional, as proposed in Pham et al. (2011), is:

$$\Theta_\varepsilon(\mathbf{u}, s) = \int_{\Omega} \left(\varphi(s, \nabla \mathbf{u}) + w(s) + \frac{1}{2} w_1 l \nabla s \cdot \nabla s \right) dx \quad (2)$$

659 where the first term in the integral represents the elastic strain energy, which is an increasing
660 function of both s and \mathbf{u} . The second term is the damage energy, a decreasing function of s , and the
661 third term is a non-local damage contribution, with $w_1 = w(1)$ and l an intrinsic material length scale
662 related to the width of the damaged regions. In the minimization of Eq. (2) is engaged a competition
663 between the first integral, which is minimized for fixed \mathbf{u} by $s = 0$, and the second one, minimized
664 by $s = 1$. But the transition from $s = 0$ to $s = 1$ is associated with a non-null value of ∇s , indeed
665 penalized in the second integral. Here, we assume:

$$\varphi(s, \text{sym} \nabla \mathbf{u}) = \frac{1}{2} s^2 \mathbf{C}[\text{sym}(\nabla \mathbf{u}^+)] \cdot \text{sym}(\nabla \mathbf{u}^+) + \frac{1}{2} \mathbf{C}[\text{sym}(\nabla \mathbf{u}^-)] \cdot \text{sym}(\nabla \mathbf{u}^-), \quad (3)$$

666 where the displacement gradient is $\nabla \mathbf{u}^+$, if $\text{div} \mathbf{u} > 0$ and $\nabla \mathbf{u}^-$, if $\text{div} \mathbf{u} \leq 0$. \mathbf{C} is the elasticity
667 tensor of an isotropic linear elastic material. Since s affects the elastic energy only when the volume
668 increases, the material experiences damage only in the case of tensile loadings, maintaining
669 undamaged in the case of compressive loadings. The damage energy density is the linear function

$$w(s) = w_1(1 - s), \quad (4)$$

670 where w_1 is related to the energy toughness G_c through the relation $w_1 = 3G_c / (4l\sqrt{2})$, deduced in
671 (Pham et al. 2011). The expression relating the internal length l and the width D of the so-called
672 process zones, i.e., that thin strip where the transition from $s = 0$ to $s = 1$ occurs, is $l = 2D / \sqrt{2}$
673 (see Pham et al., 2011 for details). In such a way, all the constitutive parameters are related to
674 quantities, which can be easily measured from experiments.

675 The model based on the energy in Eq. (2) mainly reproduces the evolution of cleavage fracture
676 (mode I fracture) due to tensile loadings, typical of brittle materials. However, in the last years,
677 many variations of the functional reported in Eq. (2) were proposed to capture different fracture
678 mechanisms. In Del Piero et al. (2007), the problem was reformulated in the most general finite
679 elasticity setting. This extension was justified by the fact that, usually, fracture is preceded by large
680 deformations. It has permitted to avoid problems of material interpenetration, and has furnished a
681 more realistic description of fracture caused by compressive loads. In Lancioni and Royer-Carfagni
682 (2009), a decomposition of the bulk energy into deviatoric and spheric parts has allowed to
683 reproduce the formation of shear bands, and their coalescence in mode II cracks, typical of quasi-
684 brittle materials. The behaviours under tensile and compressive loads have been differentiated in
685 Freddi and Royer-Carfagni (2010).

686 The functional in Eq. (2) is numerically minimized by means of incremental energy minimization, a
687 powerful mathematical tool used in many problems of fracture and plasticity (Del Piero et al. 2013;
688 Lancioni 2015; Lancioni et al. 2015). At each time increment, an iterative scheme is performed,
689 consisting in finding the local minimizers of the functional in Eq. (2), keeping s and \mathbf{u} fixed,
690 respectively. Since the functional reported in Eq. (2) is quadratic with respect to \mathbf{u} and s , separately,
691 the numerical algorithm is based on a sequential quadratic programming scheme.

692 Herein simulations are performed under the hypothesis of generalized plane stress. The geometric
693 and material characteristics are assumed according to the values reported in Table 1. Further quantities
694 required by the variational model are the fracture toughness $G_c=0.039$ N/mm and the internal length
695 $l=750$ mm. The numerical code includes a mesh refinement, which operates in those parts of the
696 body where cracks develop. In these parts the refined mesh size is $\Delta x=100$ mm. The loading step is
697 $\Delta q = 10^{-5}$ N/mm².

698 The obtained ductility is equal to 2.94 for the $[10 \times 40]$ beam, and 3.12 for the $[4 \times 40]$ one.
699 The fracture field s for different values of the load q are plotted in Figure 16 and Figure 17 for the
700 $[10 \times 40]$ and $[4 \times 40]$ beams, respectively. The evolution of the fracture is similar in the two
701 simulations. The fracture forms in the left bottom corner and evolves horizontally toward the central
702 part, at the base of the wall. Then multiple cracks form on the left side and develop toward the
703 centre of the beam base. It has been observed that, in the case of the $[4 \times 40]$ beam, the model gives
704 a diffuse representation of the fracture, and it does not distinguish single fractures, because of the
705 small width of the beam, comparable with the width of the damage localization zone.

706

707 ***Synthesis of the results and comparison***

708 The analysis of the pushover curves allow to make comparison among them, and the following
709 parameters were specifically estimated: i) the initial elastic stiffness (k_e); ii) the maximum base
710 shear (V_{max}) and the correspondent load peak multiplier ($\alpha_{max}=V_{max}/W$); iii) the secant stiffness at
711 60% of maximum load (k_s); iv) the ultimate base shear (V_u) and the ultimate load multiplier
712 ($\alpha_u=V_u/W$); v) the elastic displacement (d_e) and the elastic drift ($\theta_e=d_e/H$); vi) the ultimate
713 displacement (d_u) and the corresponding ultimate drift ($\theta_u=d_u/H$); vii) the ductility ($\mu=d_u/d_e$). In
714 addition, when the code allows for this calculation, the fundamental period (T_1) was calculated.
715 The initial elastic stiffness, the secant stiffness, the maximum base shear and the ultimate
716 displacement can be directly evaluated from the pushover diagram. The ultimate displacement has
717 been estimated as the displacement corresponding to a shear value equal to $0.85 \cdot V_{max}$ in the post-
718 peak softening branch of the pushover curve; for those models where damage has not been taken
719 into account, the maximum reached displacement has been considered, *i.e.* the one just before the
720 absence of convergence in the analysis. The ultimate load multiplier, the elastic drift and the

721 ductility were evaluated, when possible, through the estimation of the equivalent bilinear system.
722 The bilinear system was evaluated according to an equivalent energy criterion and taking into
723 account an initial elastic branch with a stiffness equal to the secant stiffness corresponding to a load
724 level equal to $0.60 \cdot V_{\max}$. For each quantity the mean value and the coefficient of variation (ratio
725 between the standard deviation and the mean value) have been calculated (Table 5 and Table 6).

726

727 ***Comparison of the pushover curves***

728 Results of both case studies ($[10 \times 40]$ in Table 5 and $[4 \times 40]$ in Table 6), show a good agreement
729 with respect to the evaluation of the initial elastic stiffness, the fundamental period, and the ultimate
730 base shear (the coefficient of variation is lower than 10%). On the contrary, differences are visible
731 with respect to the estimation of the ultimate displacement (and drift, and consequently ductility).
732 Comparisons of the pushover curves are reported in Figure 18 and Figure 19. As general remark, all
733 the codes agree with respect to the evaluation of the stress conditions. With respect to the evaluation
734 of the ultimate displacement, the numerical instruments can be grouped as follows:

735

- 736 a) codes 3Muri and 3DMacro evaluate the collapse load in accordance with the Italian
737 recommendation (DM2008, 2008) based on a selected ultimate drift;
- 738 b) codes DIANA, Code ASTER, VDM, and SMARTmasonry (with damage) provide an estimation
739 of the ultimate displacement according to the adopted damage models that allows for the
740 evaluation of the softening branch of the pushover curve; this estimation depends on the
741 parameters ruling the evolution of the damage (*e.g.* fracture energy);
- 742 c) the code ANSYS performs control force analysis hence is able to reproduce the initial branch of
743 the pushover curve, but not the descending branch; then the ultimate drift is the one
744 corresponding to the maximum base shear;
- 745 d) codes NOSA-ITACA and MADY do not have control on the ultimate displacement, which can
746 be carried on until numeric convergence is reached. The analyses shown in the paper have been
747 performed in force control mode and carried on until numerical stability of the results is
748 guaranteed;
- 749 e) the code SMARTmasonry (without damage) originates a capacity curve where the reduction of
750 strength of 85% is reached due to geometrical effects at very large values of the drift.

751

752 All the codes, but especially those adopting a damage model, provide estimations of the ultimate
753 load quite different one from each other, highlighting a very strong dependence of the collapse
754 displacement on the adopted constitutive model. The issue of the displacement capacity for masonry
755 panels with dependence on their slenderness and compressive state is addressed by Orlando et al.

756 2016. This is a very critical point since, for instance, the Performance-Based Earthquake
 757 Engineering (PBEE) or the Capacity Spectrum Method (CSM) take into account displacements as
 758 verification parameters.

759

760 ***Further discussion and additional analyses***

761 The results showed a significant dispersion associated with deformation capacity and post-peak
 762 softening branch of the pushover curves. This dispersion can significantly affect the predicted
 763 collapse performance affecting, on its turn, the confidence in structural analysis results. This
 764 demonstrated that model framework uncertainties (the uncertainties that are due to the uncertainty
 765 in the underlying science and algorithms of a model) play a key role when employed to assess the
 766 nonlinear behavior of masonry structures.

767 Starting from the available results, some additional analyses were performed in order to better
 768 characterize the ultimate displacement of the two examined beams and to estimate the reliability
 769 level connected to the epistemic uncertainty.

770 Firstly, a mean collapse curve has been determined, by considering the i -th average load multiplier
 771 $\alpha_{i,avg}$ obtained among the m_i analyses still running at each of the i -th displacement level d_i , *i.e.*

$$\alpha_{i,avg} = \frac{1}{m_i} \sum_{j=1}^{m_i} \alpha_j(d_i) \quad (5)$$

772 with:

$$m_i = \sum_{j=1}^n \delta_{ij}, \quad \delta_{ij} = \begin{cases} 1 & \text{if } d_{max,j} \geq d_i \\ 0 & \text{if } d_{max,j} < d_i \end{cases} \quad (6)$$

773 where $n=10$ (in case of SMARTmasonry, only the case with damage has been considered) is the
 774 number of the approaches, $d_{max,j}$ the maximum displacement estimated for the j -th approach and
 775 $\alpha_j(d_i)$ the load multiplier of the j -th approaches at the i -th displacement d_i .

776 As, for a given force value, each analysis provides different values of the displacement d (some
 777 analyses are force controlled, so that no fixed values for displacements are assumed), all curves
 778 have been resampled at given values of displacements d_i ; the spacing $\Delta_i = d_{i+1} - d_i$, $i = 1, \dots, N$
 779 has been selected as 1/100 of the maximum displacement obtained among all curves ($N=100$).

780 At the same time, a curve reporting for each displacement level d_i the value of the number m_i of
 781 analyses still considering an ultimate displacement $d_{max,j}$ higher than the considered displacement
 782 level d_i has been evaluated. Once the obtained histogram has been normalized with respect to the
 783 total number n of the performed analysis, it can be fitted in a way of representing an exceedance
 784 probability distribution of the d_i values (reliability curve); the obtained curve then represents the
 785 reliability level of the displacement level d_i . In other words, if m_i analyses out of n assert that this

786 displacement can be reached and if the same degree of soundness is given to each individual
787 analysis, it means that this value can be viewed as the value which has a probability

$$Prob[d_{max} \geq d_i] = \frac{m_i}{n} \quad (7)$$

788 of being exceeded. The value corresponding to an exceedance probability of 0.50 (median) has been
789 selected as the most probable collapse value for the examined structures. As shown in Figure 20, the
790 two examined cases exhibit a similar behavior and the proposed curve can be retained as a
791 reliability function for the ultimate displacement; the spreadiness of the distribution makes evidence
792 of the large variability of this parameter.

793 A second analysis has been performed by analyzing the standard deviation values σ_i at each d_i level;
794 the parameter accounts for the dispersion of the results given by the various analyses at an assigned
795 displacement level, and it is meaningful only if $m_i > 2$.

796 The obtained σ_i values have then been normalized with respect to mean value $\alpha_{i,avg}$ previously
797 determined, then obtaining the function describing the evolution of the Coefficient of Variation
798 (CoV) at increasing displacement levels:

$$(CoV)_i = \frac{\sigma_i}{\alpha_{i,avg}} \quad (8)$$

799 In Figure 21, the mean curve and the curves corresponding to $\alpha_{i,avg} + \sigma_i$ and $\alpha_{i,avg} - \sigma_i$ have been
800 reported. In the same graphs, the values of $(CoV)_i$ as a function of d_i have been reported, too.

801 Due to the strong analogy between the graphs related to $[10 \times 40]$ and $[4 \times 40]$ beams, a
802 normalization has then been proposed, by scaling the ordinates and the abscissae of the two curves
803 with respect to the values at the elastic limit, *i.e.* to the values α_{el} and d_{el} corresponding to the end of
804 an equivalent elastic branch. The latter has been determined, for each analysis, as the displacement
805 value at which the difference between the actual force level and the one obtained by using the
806 elastic stiffness (*i.e.* the value corresponding to the initial tangent stiffness) overpass a given
807 threshold fixed at 5%.

808 With respect to this normalization, the two curves show a very similar trend (as reported in Figure
809 22). It can be stated that both the analyses give the same normalized ultimate load level, which can
810 be estimated as $\alpha_{avg} / \alpha_{el} \cong 1.7$ (*i.e.* the ultimate load is 1.7 times the load at the elastic level),
811 while the ultimate normalized displacement, which plays the role of the ductility of the beams,
812 depends on the geometry; in the investigated cases, the average ductility (with respect to the
813 performed analyses) can be assumed equal to about $d/d_{el} \cong 6$.

814

815 It is indeed interesting to observe that, in normalized form, the evolution of the coefficient of
816 variation with respect to the displacement level has the same shape for both cases, so a common
817 expression can be proposed as a reasonable approximation of the obtained results:

$$(CoV)_i = 0.025 + 0.01 \frac{d_i}{d_{el}} \quad (9)$$

818 or, in an equivalent form:

$$\sigma_i = \sigma(V_i/V_{el}) = \frac{V_i}{V_{el}} \left[0.025 + 0.01 \frac{d_i}{d_{el}} \right] \quad (10)$$

819 The obtained line is reported as a dashed line in the graphs in Figure 22.

820

821 This result is quite relevant as it assesses that the obtained average curve has a growing standard
822 deviation (*i.e.* a lowering confidence level) which is proportional to both the load level and the
823 displacement level.

824 The same level of soundness can be attributed to the result of a single analysis to account for the
825 epistemic uncertainty. As a first approximation, the obtained value for the standard deviation could
826 be attributed to each of the curves obtained by the different analyses to take into account the
827 uncertainties related to the specific mechanical and numerical model.

828

829 **Concluding remarks**

830

831 The paper reported the results of a blind benchmark aimed at comparing the results obtained with
832 different analytical models and/or numerical analysis techniques (macro-elements, equivalent
833 frame, finite elements, energy approach, etc.) for the assessment of the structural behavior of a
834 series of slender masonry elements under increasing horizontal loads. The comparison, aimed at
835 deepening the effects of the so-called model framework uncertainties, showed a good agreement in
836 terms of all the main parameters that define the capacity curve, except for the ultimate drift, whose
837 determination is crucial to the audit: it is subject to uncertainties that are reflected on the entire
838 chain of seismic risk assessment.

839 As a matter of fact, all the approaches, but especially those adopting a damage model, provide
840 estimations of the ultimate load quite different one from each other, highlighting a very strong
841 dependence of the collapse displacement on the employed constitutive model. This is a very critical
842 point, since deformation capacity has a direct consequence in decision-making, because both the
843 assessment (or retrofitting) of existing structures and the design of new buildings depend on seismic
844 capacity predictions: for instance, the Performance-Based Earthquake Engineering (PBEE) or the
845 Capacity Spectrum Method (CSM) take into account displacements as verification parameters. This
846 enlightens that, at the present state of knowledge, engineering expert judgment still plays a strategic

847 role to assess the seismic safety of masonry structures when a nonlinear numerical code is
848 employed.

849

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851

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855

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1044 Table 1. Mechanical parameters: E_w (elastic modulus); ν (Poisson's coefficient); f_{wc} (uniaxial
1045 compressive strength); f_{wt} (uniaxial tensile strength); τ_k (characteristic shear strength).

1046

E_w (N/mm ²)	ν	f_{wc} (N/mm ²)	f_{wt} (N/mm ²)	τ_k (N/mm ²)
1500	0.25	5.00	0.24	0.293

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1049 Table 2. DIANA model: additional mechanical parameters.

1050

Poisson's coefficient ν (-)	Fracture energy G_f (N/m)	crack bandwidth h (m)	shear retention factor β (-)
0.00	38.97	0.75	0.35

1051

1052

1053 Table 3. Code ASTER model: additional mechanical parameters required to define the Mazars
1054 model.

1055

Parameter	Value
κ_0 damage threshold [-]	$3.2 \cdot 10^{-5}$
A_c shape coefficient (compression asymptote) [-]	1.0
B_c shape coefficient (compression peak) [-]	$2.2 \cdot 10^3$
A_t shape coefficient (tensile asymptote) [-]	0.8
B_t shape coefficient (tensile peak) [-]	$2 \cdot 10^4$
β coupling coefficient of the damage in compression and tension [-]	1.06

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1060 Table 4. ANSYS model: additional parameters for DP yield criterion and WW failure surface.

1061

Nonlinear Material parameters	Model 1 (DP&WW)	Model 2 (WW)
Drucker-Prager yield criterion parameters		
c (cohesion)	0.5384 MPa	/
η (flow angle)	65.69°	/
φ (friction angle)	43.79°	/
f_{cDP} (uniaxial compressive strength)	5.0 MPa	/
f_{tDP} (uniaxial tensile strength)	0.34 MPa	/
Willam-Warnke failure criterion parameters		
F_c (uniaxial compressive strength)	6 MPa	5 MPa
F_t (uniaxial tensile strength)	0.24 MPa	0.24 MPa
β_c (shear transfer coeff. for close cracks)	0.75	0.75
β_t (shear transfer coeff. for open cracks)	0.25	0.25

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1069 Table 5. Cantilever beam [10 × 40]. Fundamental period (T_1), initial elastic stiffness (k_e), peak multiplier ($\alpha_{max}=V_{max}/P$), ultimate drift ($\theta_u=d_u/h$), secant
 1070 stiffness (k_s), elastic drift ($\theta_e=d_e/h$), ductility ($\mu=d_u/d_e$), ultimate load multiplier ($\alpha_u=V_u/P$).
 1071

		3Muri	3DMacro	DIANA	Code	ANSYS		NOSA	MADY	VDM	SMARTmasonry		Mean	CoV
					ASTER	(1)	(2)				(3)	(4)		
T_1	[s]	0.99	-	1.13	1.13	1.14	1.14	1.33	1.09	-	1.14	1.14	1.14	0.0768
k_e	[kN/mm]	34.78	36.50	37.22	36.80	37.10	37.23	36.79	39.06	37.00	36.32	36.32	36.83	0.0275
α_{max}	[-]	0.193	0.212	0.195	0.212	0.208	0.199	0.178	0.217	0.203	0.206	0.206	0.203	0.0542
θ_u	[‰]	8.30	4.03	8.30	14.10	10.63	6.90	11.31	18.46	5.00	8.60	51.19	13.35	0.9888
k_s	[kN/mm]	33.52	35.90	37.15	-	-	-	35.11	-	-	30.66	30.65	33.83	0.0806
θ_e	[‰]	2.44	2.46	2.18	1.30	5.28	1.32	2.16	1.63	1.70	2.74	2.69	2.35	0.4661
μ	[-]	3.39	1.64	3.80	10.85	8.05	5.24	5.24	11.33	2.94	3.08	19.06	6.78	0.7655
α_u	[-]	0.182	0.196	0.184	0.212	0.208	0.199	0.168	0.209	0.203	0.194	0.187	0.195	0.0692

Note: (1) DP&WW criteria; (2) WW criterion; (3) with damage; (4) without damage.

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1078 Table 6. Cantilever beam [4 × 40]. Fundamental period (T_1), initial elastic stiffness (k_e), peak multiplier ($\alpha_{max}=V_{max}/P$), ultimate drift ($\theta_u=d_u/h$), secant
 1079 stiffness (k_s), elastic drift ($\theta_e=d_e/h$), ductility ($\mu=d_u/d_e$), ultimate load multiplier ($\alpha_u=V_u/P$).

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		3Muri	3DMacro	DIANA	Code	ANSYS		NOSA	MADY	VDM	SMARTmasonry		Mean	CoV
					ASTER	(1)	(2)				(3)	(4)		
T_1	[s]	2.25	-	2.73	2.68	2.76	2.76	2.73	2.71	-	2.84	2.84	2.70	0.0659
k_e	[kN/mm]	2.38	2.43	2.48	2.24	2.48	2.48	2.48	2.50	2.53	2.29	2.29	2.42	0.0415
α_{max}	[-]	0.076	0.0756	0.084	0.071	0.082	0.077	0.073	0.087	0.088	0.068	0.068	0.077	0.0923
θ_u	[‰]	15.10	10.00	22.46	11.38	18.21	12.35	25.25	43.80	15.00	24.61	34.61	21.16	0.4968
k_s	[kN/mm]	2.31	2.43	2.47	-	-	-	-	-	-	2.13	2.13	2.29	0.0701
θ_e	[‰]	5.55	5.04	5.64	4.23	3.33	3.31	5.00	6.50	4.80	5.43	5.25	4.92	0.1973
μ	[-]	2.72	1.98	3.98	2.69	5.47	3.73	5.00	6.73	3.12	4.53	6.59	4.23	0.3759
α_u	[-]	0.071	0.076	0.079	0.071	0.082	0.077	0.068	0.083	0.080	0.065	0.063	0.074	0.0930

1081 Note: (1) DP&WW criteria; (2) WW criterion; (3) with damage; (4) without damage.

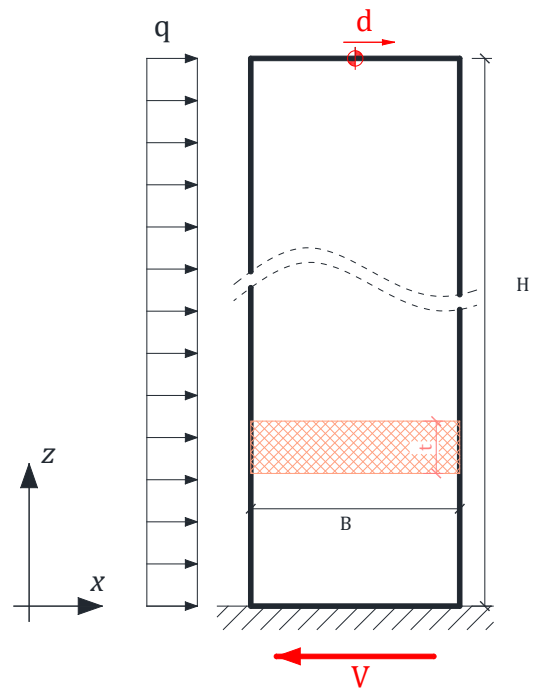
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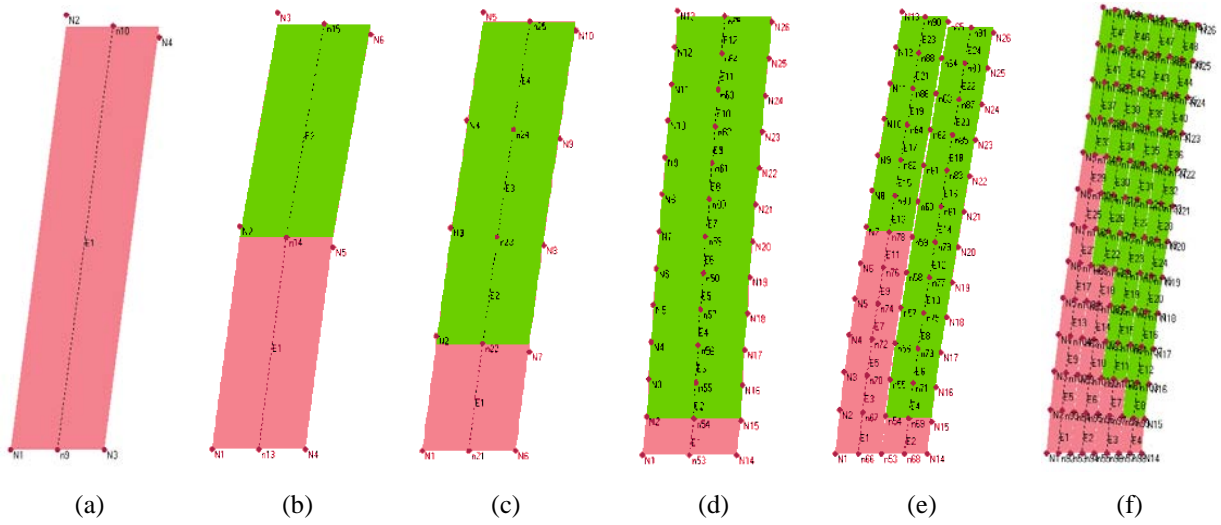
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Figure 1. The cantilever masonry beam.

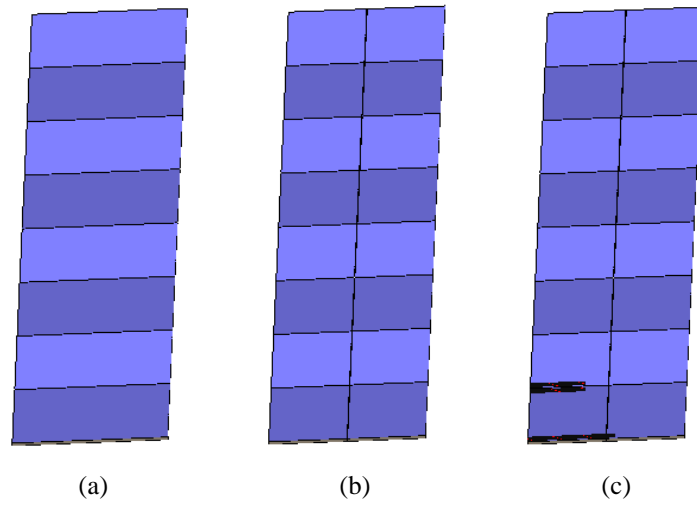
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Figure 2. 3Muri: discretization of the $[10 \times 40]$ cantilever masonry beam (pink elements denote those collapsed at the end of the pushover analysis).

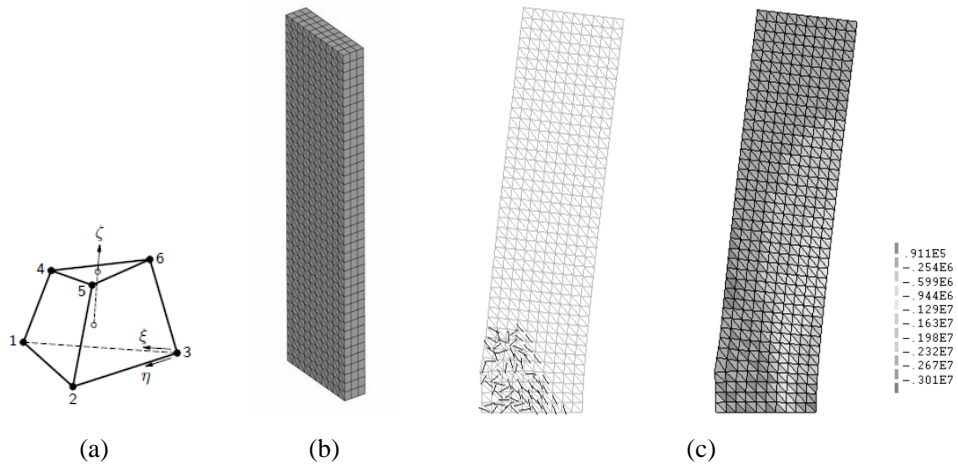
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Figure 3. 3DMacro: discretization of the $[10 \times 40]$ cantilever masonry beam and corresponding collapse configurations.

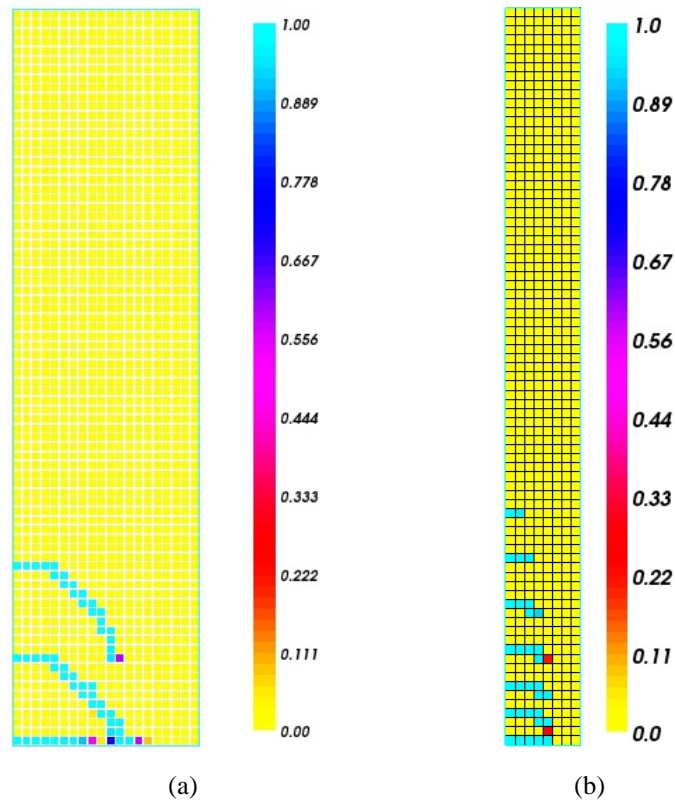
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Figure 4. DIANA: (a) 6-node isoparametric TP18L element; (b) discretization of the $[10 \times 40]$ cantilever masonry beam; (c) cracking pattern and vertical stresses corresponding to the maximum base shear for the $[10 \times 40]$ cantilever beam.

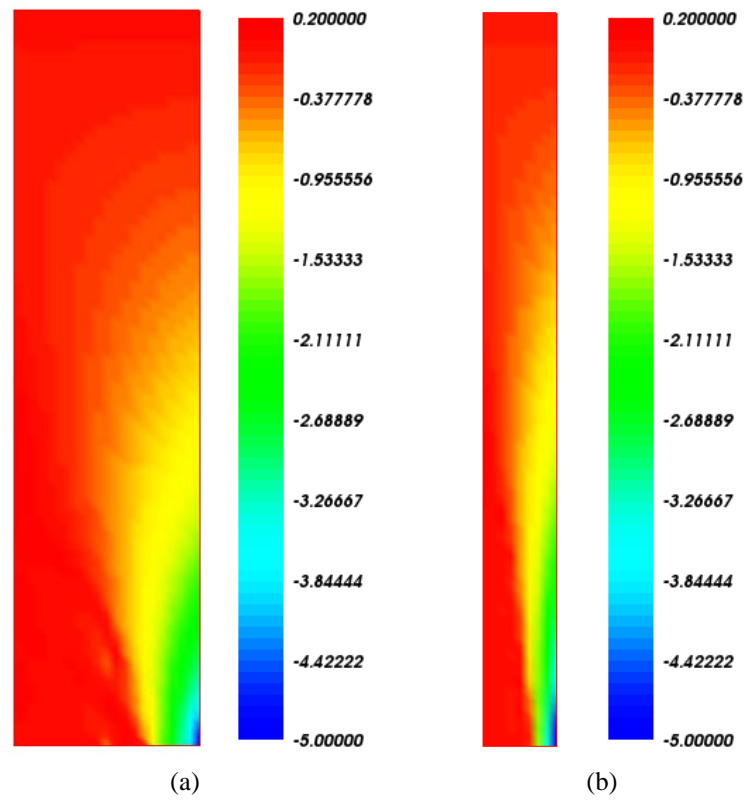
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Figure 5. Code ASTER: damage map at collapse for (a) $[10 \times 40]$ and (b) $[4 \times 40]$ cantilever masonry beams.

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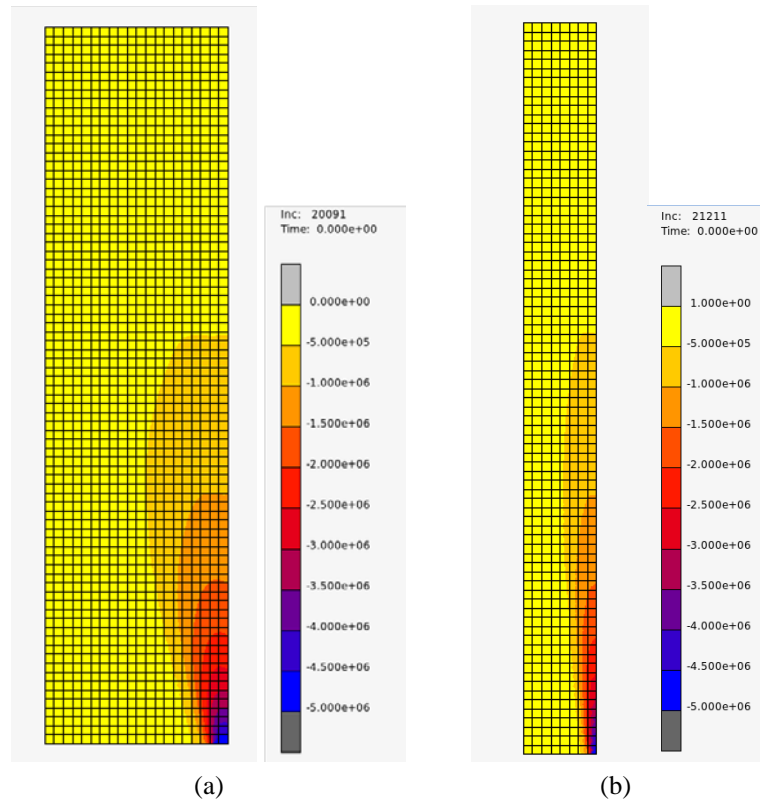
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Figure 6. Code ASTER: principal compressive stresses (MPa) at collapse for (a) $[10 \times 40]$ and (b) $[4 \times 40]$ cantilever masonry beams.

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1159 Figure 9. NOSA-ITACA: distribution of the σ_{zz} (N/m^2) stress tensor component at collapse obtained
1160 for (a) $[10 \times 40]$ and (b) $[4 \times 40]$ cantilever masonry beams.

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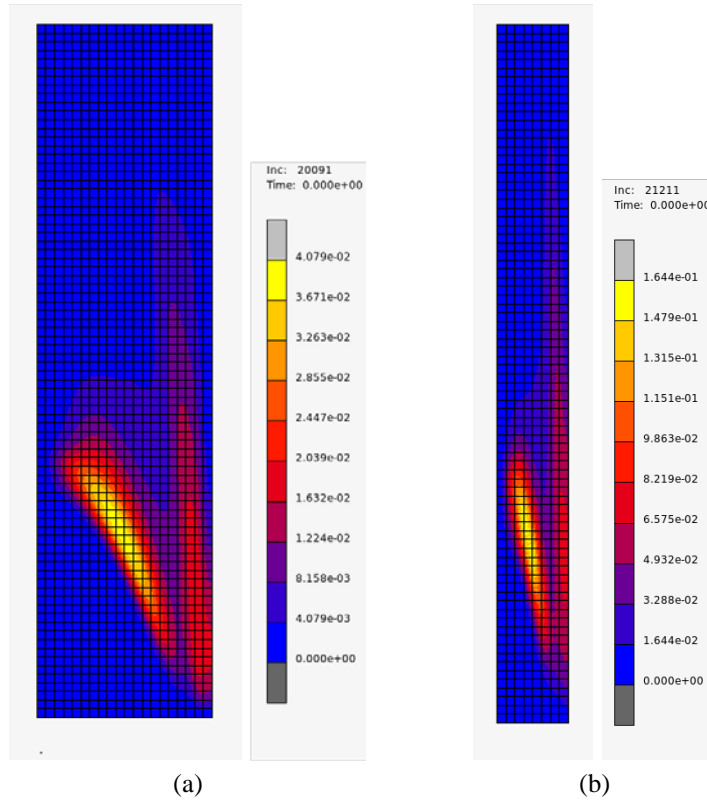
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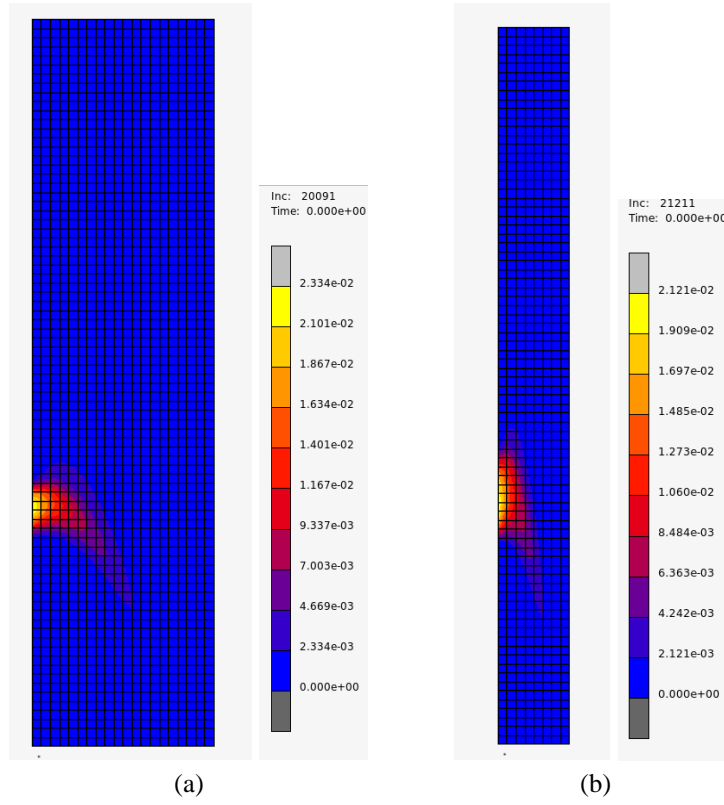
1168 Figure 10. NOSA-ITACA: distribution of the ε^f_{xx} component of the fracture strain tensor at collapse
1169 obtained for (a) $[10 \times 40]$ and (b) $[4 \times 40]$ cantilever masonry beams.

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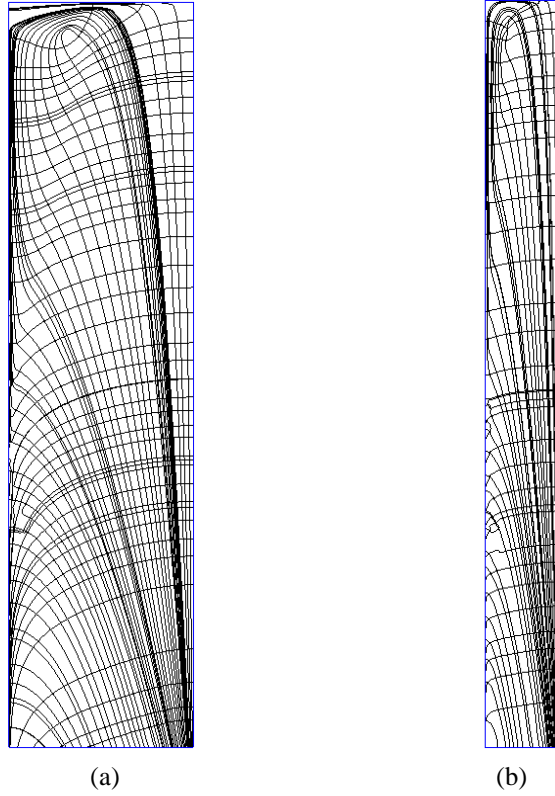
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Figure 11. NOSA-ITACA: distribution of the ε^f_{zz} component of the fracture strain tensor at collapse obtained for (a) $[10 \times 40]$ and (b) $[4 \times 40]$ cantilever masonry beams.

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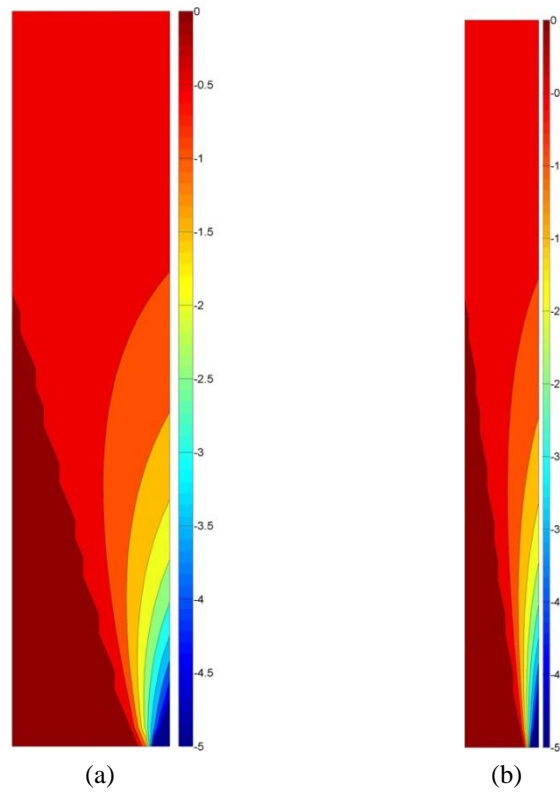
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1186 Figure 12. NOSA-ITACA: distribution of the isostatic lines at collapse obtained for (a) [10 × 40]
1187 and (b) [4 × 40] cantilever masonry beams.

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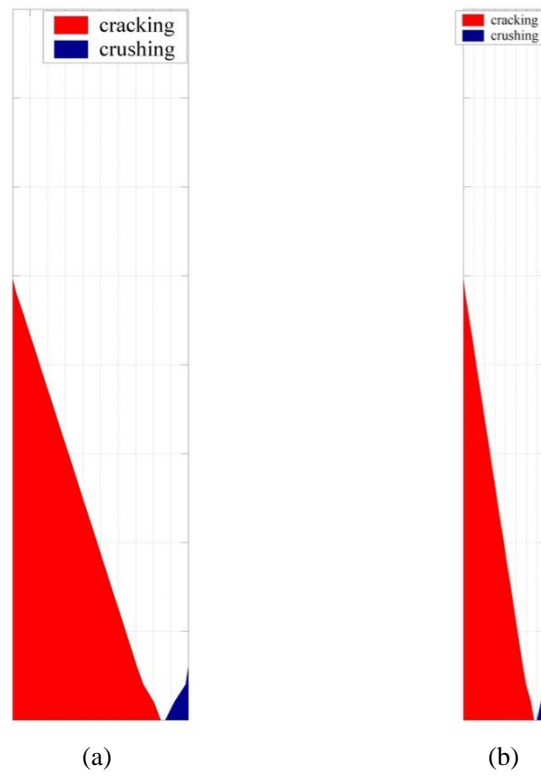
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1194 Figure 13. MADY: distribution of the axial stress σ_z (MPa) at collapse for (a) [10 × 40] and
1195 (b) [4 × 40] cantilever masonry beams.

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1203 Figure 14. MADY: distribution of the damage map at collapse for (a) $[10 \times 40]$ and (b) $[4 \times 40]$
1204 cantilever masonry beams.

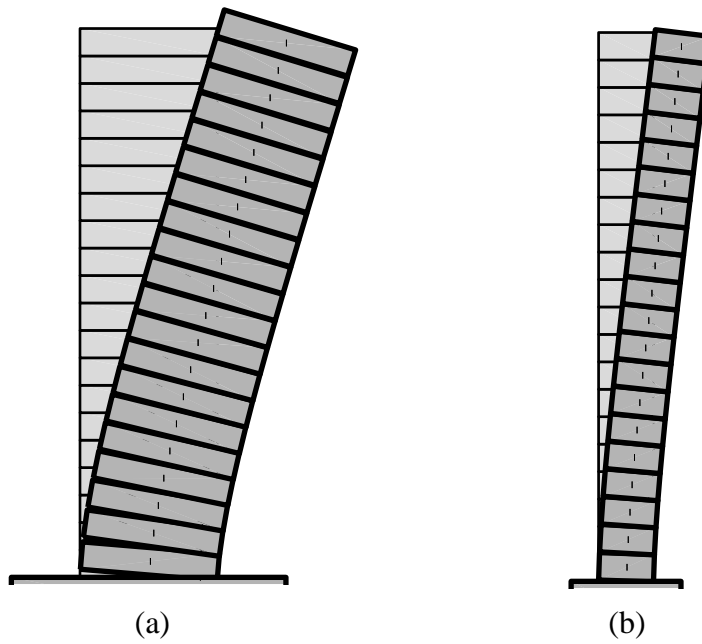
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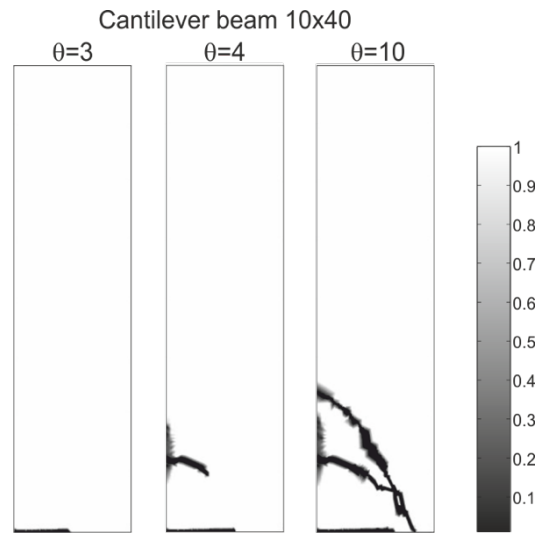
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1211 Figure 15. SMARTmasonry: deformed shape at collapse for (a) [10 × 40] (displacement
1212 amplification factor 47.0) and (b) [4 × 40] (displacement amplification factor 7.5) cantilever
1213 masonry beams.

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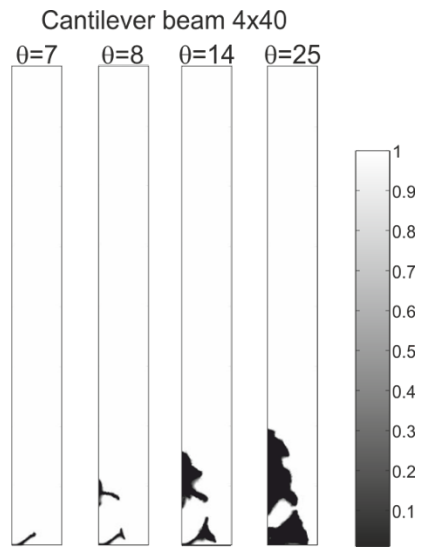
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Figure 16. VDM: damage field s at different values of the drift θ for the $[10 \times 40]$ cantilever masonry beam.

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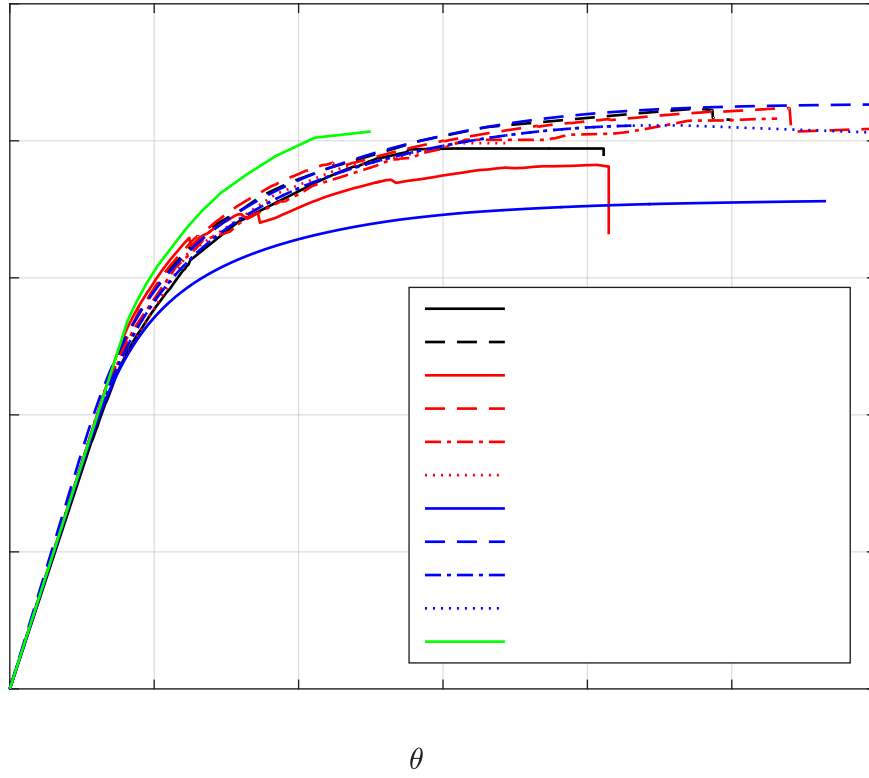
1231 Figure 17. VDM: damage field s at different values of the drift θ for the $[4 \times 40]$ cantilever
1232 masonry beam.

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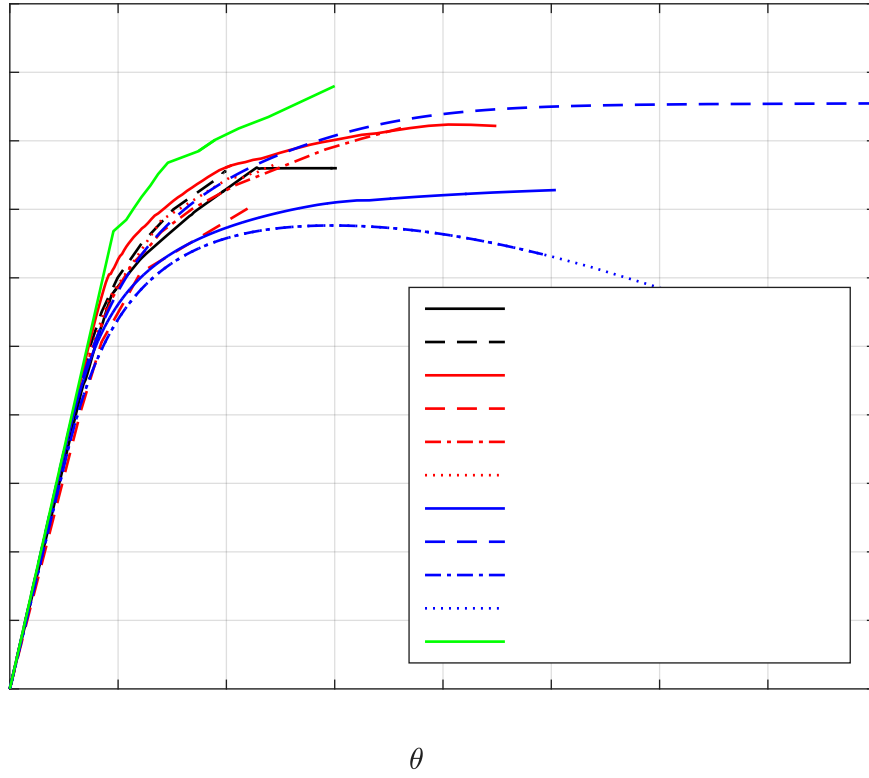
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1241 Figure 18. Comparison of pushover curves for [10 × 40] cantilever beam.

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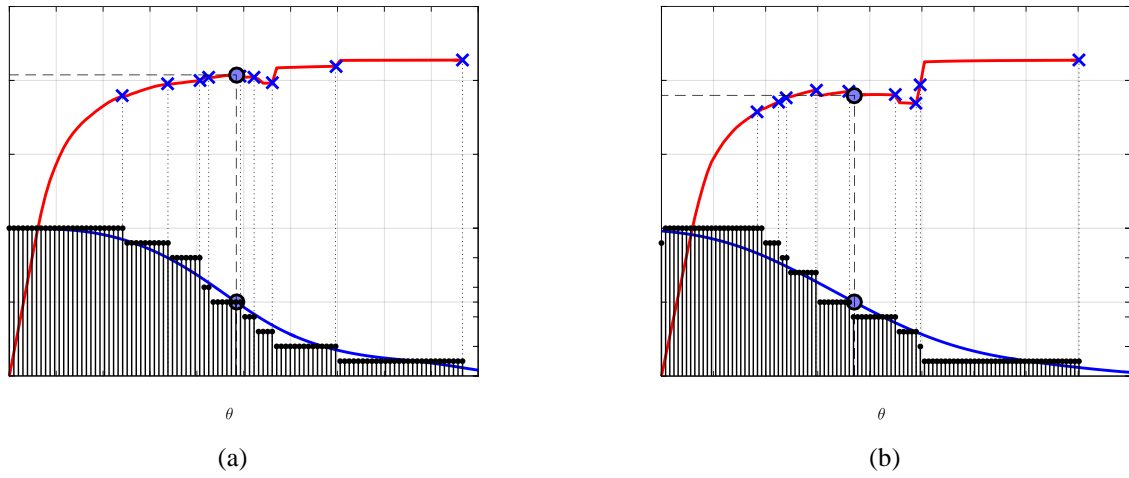
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Figure 19. Comparison of pushover curves for [4 × 40] cantilever beam.

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1257 Figure 20. Mean capacity curve for (a) [10 x 40] and (b) [4 x 40] beams. Collapse points for the
1258 various approaches are reported as crosses onto the curve, median value of the collapse
1259 displacement is reported as a heavy circle. The histogram below reports, at each displacement level,
1260 the number of analyses still active together with its curve fitting.

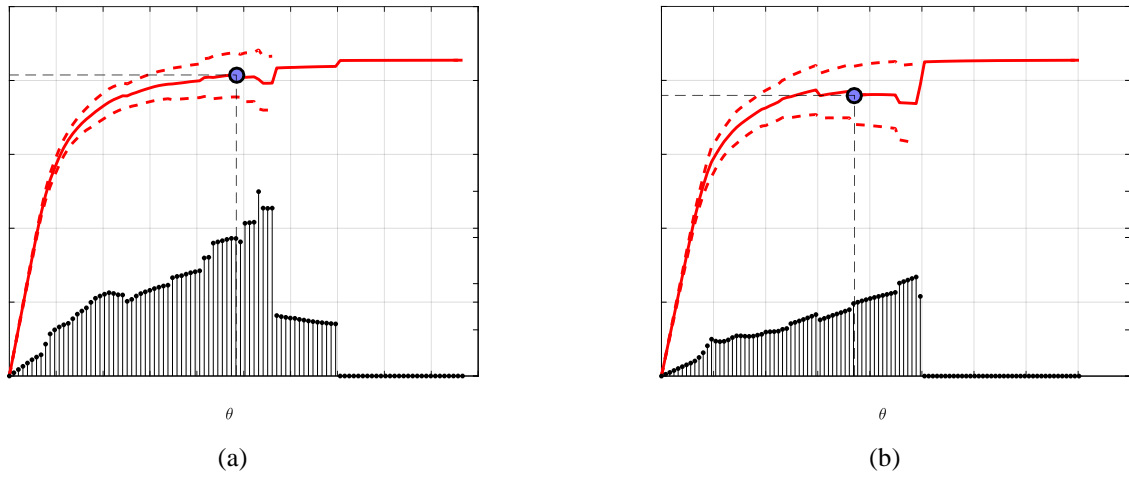
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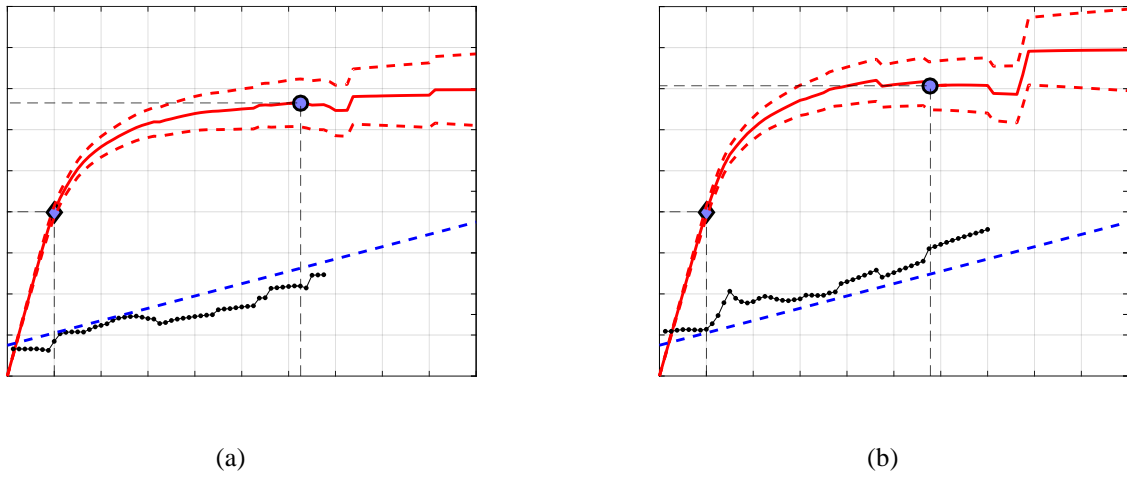
1267 Figure 21. Mean capacity curves (μ_i) for (a) [10 x 40] and (b) [4 x 40] beams. Median collapse
1268 point is reported as a solid circle onto the curve, dashed curves represent the curves $\mu_i + \sigma_i$ and $\mu_i - \sigma_i$.
1269 The histogram below reports, at each displacement level, the standard deviation of the value
1270 according to the curves, referring to still active analyses.

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1278 Figure 22. Normalized mean capacity curves (μ_i) for (a) $[10 \times 40]$ and (b) $[4 \times 40]$ beams. Median
1279 elastic-limit and collapse points are reported as solid rhombus and circle onto the curve, red dashed
1280 curves represent the normalized curves $\mu_i + \sigma_i$ and $\mu_i - \sigma_i$. The graph below reports the CoV for the
1281 nondimensional curve (blue dashed line is the approximation reported in the text).

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